

Trials Classic Flyers October 2014

Experimental 406MHz Antenna Location testing within the confines of a helicopter tail boom

The background to these tests is that over the past 4-5 years we have seen a dismal rate of activations on 406MHz ELT signals being receipted by the Rescue Coordination Centre. Typical accidents either break the external antenna and or fracture the coaxial cable and or, in particular, only of one brand, a failure of the G-switch to activate either by G-switch failure and or because of the way the ELT was / is mounted.

It is apparent, in New Zealand, that a high number of Artex ME406 (AF) style installed in helicopters are fitted either horizontal or at approximately 23 degrees rather than the previously accepted 45 degrees.

This is due to the original manufacturer manual suggesting that installers of the ELT refer to the helicopter manufacturer and the helicopter manufacturer not having any written guidelines. The early ELT manuals also referred to not installing greater than 23 degrees to avoid preloading the G-Switch. Of course all these early guidelines referred to fixed wing installations, not rotary wing.

Consequently, these installations are probably skewing the non-activation rate but at this stage, the NZ regulator has not taken any action to remedy the situation despite warnings being raised in 2012.

During the development of a secondary antenna switching device (SASD) in New Zealand in 2006, we placed a 406AP in the tail boom of a Eurocopter EC120 and tested the ELT with a test protocol with homing signal. The test was receipted by the Cospas-sarsat satellites and the homing signal, although weak because of the metal structure, was radio direction located from over a mile away at only 1200 feet altitude. (See attached report from 2006 and further actual testing in February 2009 of the SASD).

Although the SASD clearly had merit and was proven to work by simply measuring the SWR of the external antenna, (before switching to the secondary antenna), we were unable to bring it to the market. The regulator, after consulting with FAA, would not grant approval to use the SASD within an existing approved TSO126 system. That refusal occurred even though the SASD did not interfere or change the "approved system" unless the "approved system" failed by way of a broken external antenna or a detached coaxial cable. We still see merit in the device especially if sold by an ELT manufacturer as it could be fitted in the aftermarket area into existing systems. Given that a secondary antenna worked on MH317, the the interest in internal secondary antennas will become very topical.

The SASD successful testing led to the Eurocopter development (in conjunction with Kannad) of the Integra ELT that has a secondary 406MHz antenna within the ELT and, in addition, a GPS. This improved development is a great leap forward to improving ELT activations within Aviation, in particular the rotary wing industry.



For an ELT to function as required, the ELT system including the antenna must be in the maximum crash worth position. To get the antenna into a crash worthy position, I believe that we should remove the antenna from the outside of the aircraft to the inside or build it into the aircraft structure. Having an ELT inside the Helicopter and the antenna on the Tail Boom is not sensible.

We could do away with the coaxial cable by using the 406AP ELT with the antenna directly attached as these tests successfully showed.

Another factor to improve crash worthiness is that when a coaxial Cable is utilised, it should have a sacrificial loop to act as a shock absorber and remove any tension from the cable and most importantly, the BNC fittings during an accident. Disconnection of the BNC fitting to the coaxial cable happens far too often.

The following test and results undertaken in Tauranga, New Zealand on the weekend of the 10th of October 2014 prove without question that the 406MHz signal from the antenna within the confines of a (non-carbon fibre) airframe will function as required.

In these tests we used different Hex ID codes over an extended period.

The first test was just a datum of a 406AP ELT on the ground to determine the power level receipted by the GEO and LEO satellites. The second and third subsequent tests were the antenna vertical within the tail boom for 12 hours then the antenna horizontal within the tail boom for 24 hours.

The result was that the signals were receipted by the GEO and LEO satellites with little loss of receipted power (dBm)

Lloyd Klee



Test 1

ELT with integrated antenna located outside in clear space with antenna vertical to ascertain a benchmark.



Dongle TEST 01 FFFED0600F2490E1A42E869ECD10 15 Hex ID C01E4921C3485D0

(Check BT100 214-452

On approx. 1508 NZLT or 0208 GMT 10 October 2014

Off 1545 NZLT or 0245 GMT 10 October 2014



Test 2
ELT with integrated antenna located inside Tail boom with antenna vertical

Dongle test 02 FFFED0600F2490E1A42C806AD810

15 Hex. ID C01E4921C348590

(Check BT100 214-451

ON 1450hours NZLT or 0150 or GMT 10 October 2014 OFF 0800 NZLT 0800 11 October or 1900 GMT 10 October 2014







Test 3
ELT with integrated antenna located inside tail boom with antenna horizontal lengthwise within





Dongle Test 03 FFFE2F600F2490E1A42805898F90

15 Hex ID C0E44921C348500

(Check BT100 214-454

Internal antenna test 2014-455 under tail boom power measured 89% Internal antenna test 2014-456 in front of tail boom power measured 74%



(in this test the BT100 was at the opposite end of the antenna tip)

ON 0810 hours 11 October NZLT or 1910 GMT 10 October 2014 OFF 0710 hours 12 October NZLT or 1810 GMT 11 October 2014

Data from the satellites was collected by John Asson (Australia)

Test 02 (Vertical) 406MHz transmission test was for approx.12 hours

Test 03 (Horizontal) 406MHz transmission test was for approx. 24 hours There was no 121.5MHz Homing signal transmitted.

The airframe had most holes that were within the immediate vicinity of the operational antenna blocked off. Antenna transmitted through the aluminium structure successfully



Tail Boom Classic Flyers 37.668 South and 176.198 East







Holes in top of tail boom were covered with aluminium tape.
The closest inspection hole to the camera was covered
Tests were above the port lowest (closest to fence)

Testing underway

BT100 photos provided for evidence of timing and hex codes





Datum Test 01 35.1 dBm 2014-451

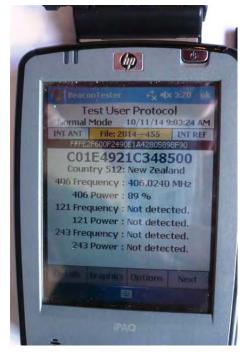
Datum Test 01 2014-452



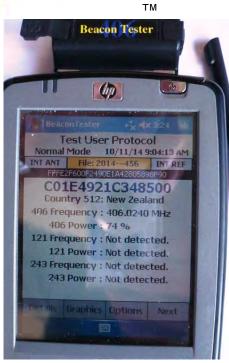
Test 02 2014-453



Test 03 2014-454



Test 03 2014-455 = 89% Power Test taken underneath tail boom



Test 03 2014-456 = 74% Power Test taken in front of tail boom





Radiated power Test 03 2014- 453 = 74% Power outside of tail boom





Port covered during TEST 02 and most of Test 03



Duct tape came undone sometime during 24 hour test of Test 03 (as it is on the underside of antenna unlikely to influence test)



ТМ



Test 02 Vertical within tail boom



TEST 03 Horizontal antenna within tail boom



Unused inspection Port covered



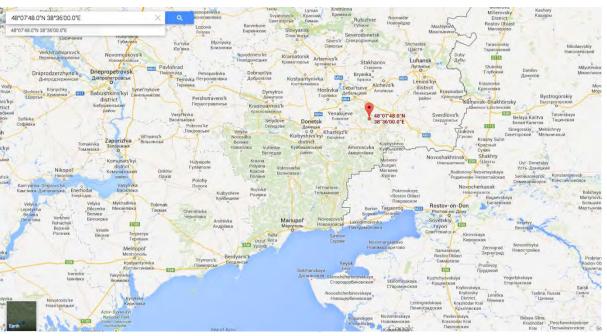
Test 01 Datum test. Power is only 35.1dBm



MH17 Crash Secondary Antenna Results

Output time : 18-JUL-2014 01:05:33.72

Alert	Sat	Sr	С	Beacor	ı ID	L	at A	Long		Lat	L	ong			
В	TCA	4	Trar	nsmit	S	ΙΤ	Route	М̂sg	Com	ms					
Key	, ID	I	D			Dį	Min	Dg	Dg	Min	Dg	Min	DAY HE	R:MN	DAY
HR:MN	I	l mi	t	Name	Nun	Н	ldlr	Sta							
19075	G21		C2A4	39E7AB	25CD′								191 1	13:32	191
13:46	ÎSÎT1	22	SI	MCC 08	628	NUL	L_IO	Tx							
19083	S11		C2A4	39E7AB	25CD′								191 1	13:32	191
13:56	์ ริเรา	22	SI	MCC 08	630	NUL	L_IO	Tx							
20803	G63		C2A4	39E7AB	25CD ⁻								198 1	13:20	198
13:35	^ŜĨŤ1	22	SI	MCC 08	745	NUL	L_IO	Tx							
20804	G62		C2A4	39E7AB	25CD′								198 1	13:20	198
13:35	ÎSÎT1	22	SI	MCC 08	746	NUL	L_IO	Tx							
20807			C2A4	39E7AB	25CD ⁻	48	8.6N	38 37.7	'E 4	4 27.51	N 1	19 50.4E	198 1	13:37	198
13:50	1 SIT1	33	SI	MCC 08	747	NUL	L_IO	Tx							
20817	S07		C2A4	39E7AB	25CD′	48	7.8N	38 36.0	E 4	8 27.91	N 4	10 23.1E	E 198 1	13:50	198
14:50	SÎT1	27		CMC 4	3783	NUL	L_IO	Tx							
20818	S07		C2A4	39E7AB	25CD′	48	7.8N	38 36.0	E 4	8 27.91	N 4	10 23.1E	198 1	13:50	198
14:50	SÎT1	27	F	FMCC 50	560 F	FTP_F	МСС	Tx							



Auxiliary Antenna Trials, 15 May 2006





The Westpac Rescue Helicopter, a BK117, undertook the RDF trials. Photo shows Darryl on left and his crew Leon and paramedic Bruce. On the extreme left, Tom McCready CAA accident investigator and trial coordinator, and on right Murray Benns, Eurocopter NZ who loaned the EC120 to use for the trial.

The tight space in tail rotor boom meant that that the auxiliary antenna on the Kannad 406MHz ELT needed to be bent through 90 degrees to avoid contact with fuselage



Inside Tail Rotor boom, structure is Aluminium and Nomex; the antenna is visible behind battery box



Photo inside Tail boom

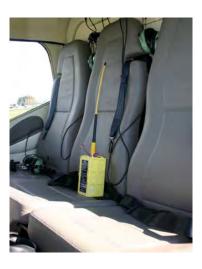


Interestingly, a satellite pass detected the 406MHz and the 243MHz but not 121.5MHz in this second serial

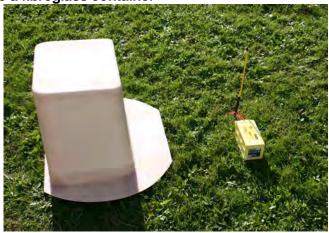
External showing tail boom access



Third serial, the cabin is composite structure including carbon fibre cloth and Nomex



Fourth trial inside a fibreglass container



Auxiliary Antenna Testing 15 May 2006 (preliminary)

Enclosure / hindrance	Time on NZLT		Radio Distance in nautical Miles at 1200 feet	RDF From at 1200 feet altitude	406MHz receipt on satellite	Satellite	121.5MHz Yes / No	243MHz Yes / No	Satellite GPS position Actual GPS position 37 02S & 174 58 E
Nil	0835	0900	12.0	12.0	Y	G09 & G10 S06	Y	Y	174 57E / 37.01S
Helicopter tail boom, Alu. & Nomex	0935	0947	4.8	1.1	Y	S09 & G10	N	Y	174 57E / 37.02S
Cockpit of EC120	0950	1000	10.4	9.4	Y	G10	N/A	N/A	
Fibreglass housing	1003	1010	12.0	7.7	Y	G10	N/A	N/A	
Notes									
а			When ELT antenr	na enclosed, RDI	Fhad little or no Au	udio			
b			Weather	was fine, all seria	ls at 1200feet				
С			ELT was a kannad	406 with 406, 12	1.5 and 243MHz o	output			
d			Test Loc	ation was 37 025	S & 174 58 E				



ACTUAL TEST of SASD 12 February 2009

Objective

The objective was a live test of the SASD utilizing the ERCEG ELT and an actual working sample SASD with two styles of secondary antennas.

General

This new test was conducted between 1000 hours and 1200 on the 12/02/2009 NZLT or 2100-2300 UTC and some will be aimed to coincide with satellite passes. We will phone at the start of each test that will be 15 minutes long, switched on 5 minutes before any specific satellite pass is due

All times are NZ standard times.

The set up was as per test as per test report No 60111.1 conducted 2 February 2006 but with the SASD between the ELT and the Whip

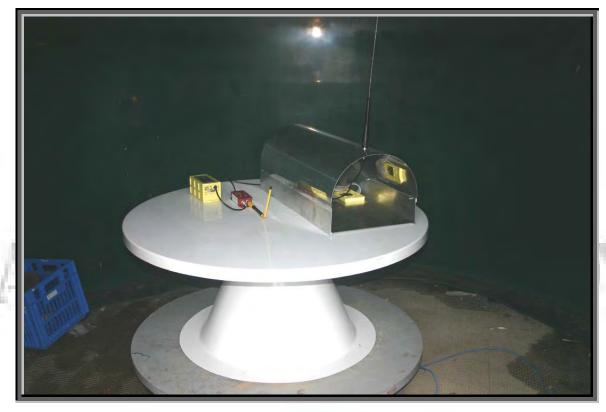
- ➤ TEST # 1 and 1A (IESM134) (Hex # C01ECC349DD40A1) Transmission of the ELT activated at 10am for 13 minutes (ANT200) then the Primary Whip antenna was disconnected so that the Secondary Antenna Switching Device was activated. (SASD) at 1013. The secondary antenna for test was a yellow (ANT110) This test duplicated the same ELT and antenna as tested February 2006. EMC conducted two power measurements during this test. This test was an open test that finished at 1026Hrs. This test was an open test
- > TEST # 2 (DONGLE KTEST 01) (C01FE870D2135D1) At 1030 the primary whip antenna was connected and the secondary antenna was replaced with a black ¼ wave antenna. The SASD was connected at 1035. This test ran until 1049. This test was an open test
- ➤ Test #3 (DONGLE KTEST 03) (C01FE870D213501) Note change of HEX Code than that originally advised. The enclosed SASD was fitted the ¼ wave stubby antenna enclosed in a structure. A minor issue occurred at 1100 when it was realized that the cable to the Primary antenna was loose. The test was restarted at 1105 on the primary antenna then SASD was switched at 1114. The enclosed SASD was switched off at 1135. This was an enclosed test.



- > Test # 4 (DONGLE KTEST 03) (C01FE870D2135D1) Note change of HEX Code than that originally advised. Enclosed SASD in structure with the ANT110 antenna connected. Note that the antenna was horizontal then bent thru 90 degrees to fit enclosure. Primary Antenna transmitted at 1140 and SASD was switched at 1145. The test finished at 1200. This was an enclosed test.
- > CONCLUSION Tests show that an enclosed SASD as tested will provide a satellite alert on both the GEO and LEO Cospas-sarsat satellites











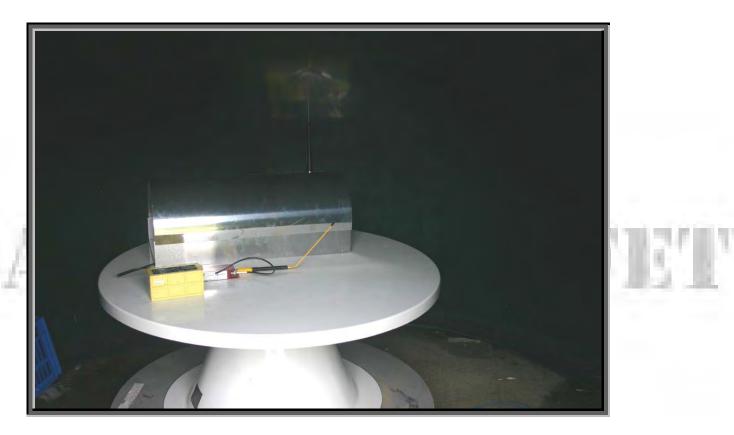


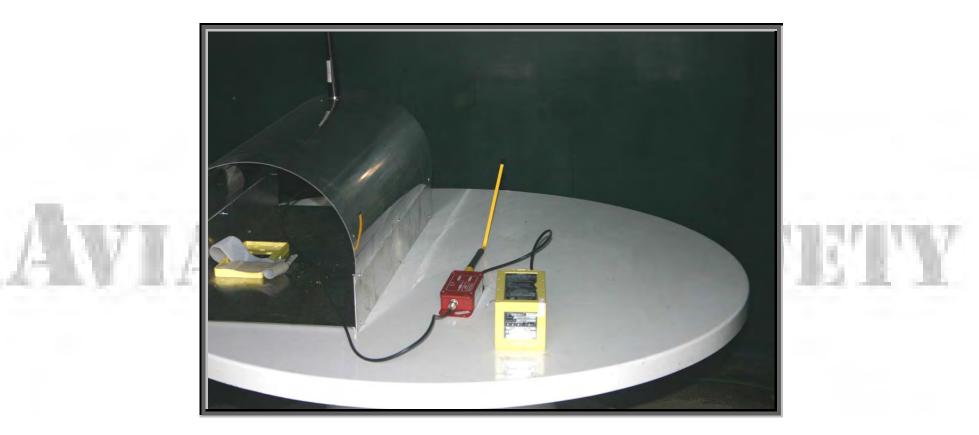
Table rotated during all power measurements (NZ local time 1004Hrs)





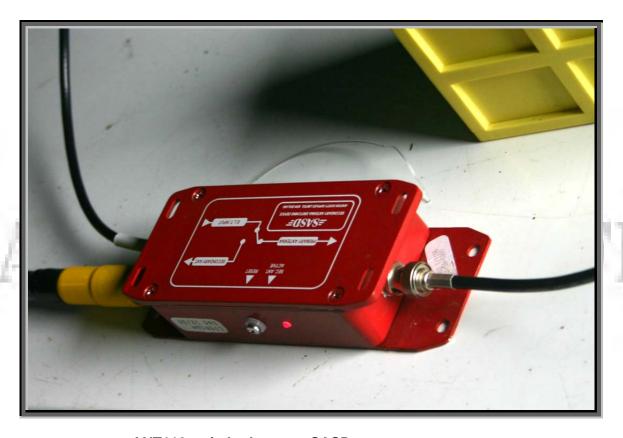
Test setup, table inside green tank, power receiver to right (NZ local time 1005Hrs)





First test 12/02/09 ANT110 open test (NZ local time 1011Hrs)





ANT110 switched over to SASD (NZ local time 1117Hrs).

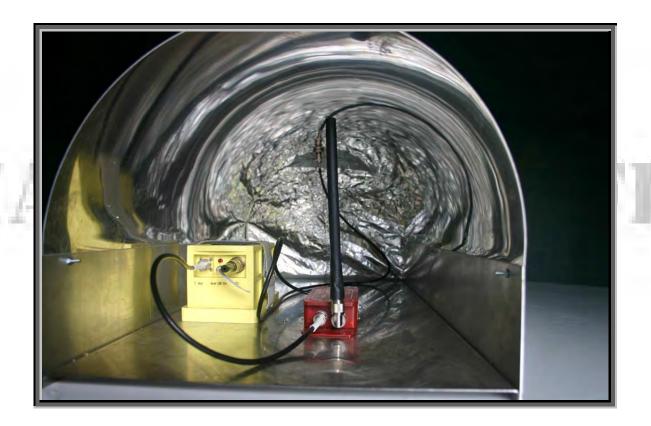
Note red LED showing SASD activated





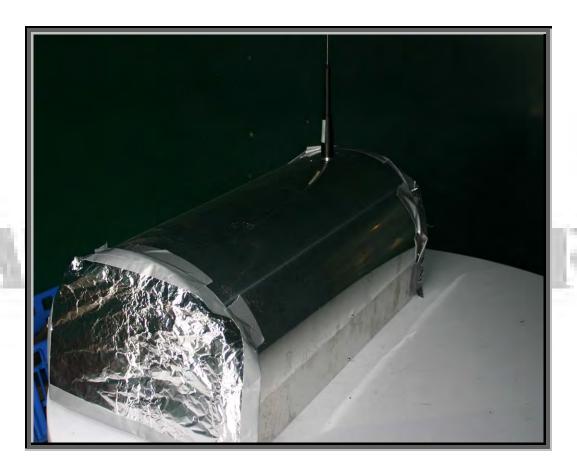
Enclosed test Enclosure rear (NZ local time 1055Hrs)





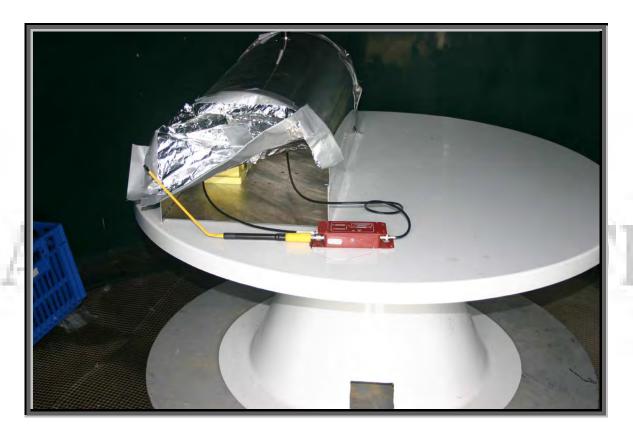
Test with 1/4 wave stubby Antenna (NZ local time 1055Hrs)





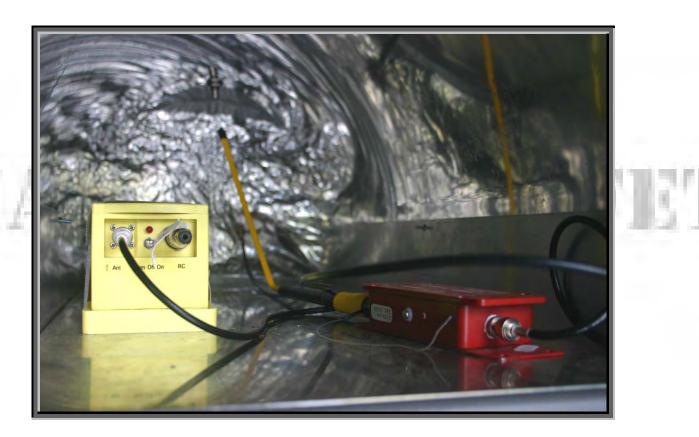
Fully Enclosed setup (NZ local time 1108Hrs)





Final test enclosed ANT110 (NZ local time 1139Hrs)





ANT110 bent to allow in test rig, note this test the 406MHz antenna is effectively horizontal



(NZ local time 1158Hrs)

GEOSTATIONARY RESULTS line 9-12 inclusive

	LUT ID	Туре	Mode	Country,	1	Beacon IC	1	Betection Time	#Msg	GPS Latitude (degN)	GPS Longitude (degE)	Satellite ID	Power (dBm)	Frequency (Hz)
7	5123	406 SARR	N	BRA	D8CC4	06630	002F1	2009/042 20:58:54	1	N/A	N/A	211	-136.6	406028091.1
	5123	406 SARR	N	BRA	D8CC4	05C6C	002F1	2009/042 18:40:55	1	N/A	N/A	211	-141.7	406028031.8
1	5123	TEST	N	CAN	A79EE	E26E3	2E1D0	2007/232 23:54:04	121	N/A	N/A	211	-135.7	406021870.9
4	5123	406 SARR	N	CAN	A78DF	01E9E	00261	2009/042 22:35:39	1	N/A	N/A	211	-142.2	406028113.3
i	5123	TEST	N	CAN	A79EE	E26E3	2E190	2009/040 15:17:47	121	N/A	N/A	211	-138.6	406022057.6
i -	5123	406 SARR	N	CHI	DAA64	BAE62	E8721	2009/042 05:52:09	4	N/A	N/A	211	-148.3	406024485.4
	5123	406 SARR	N	HON	A9C8C	70A34	D34D1	2009/042 22:24:09	121	N/A	N/A	211	-136.1	406027191.5
	5123	406 SARR	N	MAL	9F090	30C34	D34D1	2009/042 20:50:15	2	N/A	N/A	211	-141.2	406025122.7
(5123	TEST	N	NZL	C01FE	870D2	135D1	2009/042 22:41:12	3	N/A	N/A	211	-137.6	406024874.3
0	5123	TEST	N	NZL	C01FE	870D2	13501	2009/042 22:07:12	4	N/A	N/A	211	-139.7	406024876.7
1	5123	TEST	N	NZL	C01FE	870D2	13591	2009/042 21:33:08	17	N/A	N/A	211	-142.8	406024876.5
2	5123	TEST	N	NZL	C01EC	C349D	D40A1	2009/042 21:02:12	26	N/A	N/A	211	-142.6	406024876.0
3	5123	406 SARR	N	UKM	9D0E1	D2BD4	00001	2009/042 14:35:45	95	N/A	N/A	211	-135.6	406027223.5
4	5123	TEST	-M-	USA	ADDF8	E345C	D35D0	2009/042 17:12:53	121	N/A	N/A-	211	-149.0	406028091.5



LUT ID	Satellite ID	Beacon ID	NZ Local time Rx	Frequency (Hz)	Rx Power (dBm)	Bit Rate (bps)	Mean Bit Power
5121	9	C01FE 870D2 13591	2009/042 10:39:51.053	406034102.2	-127.03	2408.692	93.345
5121	9	C01FE 870D2 13591	2009/042 10:40:39.290	406033352.6	-120.26	2409.056	95.058
5121	9	C01FE 870D2 13591	2009/042 10:41:27.507	406032966.4	-123.64	2409.056	95.058
5121	9	C01FE 870D2 13591	2009/042 10:42:17.844	406032260.7	-121.95	2409.056	95.058
5121	9	C01FE 870D2 13591	2009/042 10:43:08.161	406030977.9	-123.64	2409.056	95.058
5121	9	C01FE 870D2 13591	2009/042 10:44:00.064	406028598.7	-119.70	2409.056	95.058
5121	9	C01FE 870D2 13591	2009/042 10:44:48.816	406025248.8	-119.70	2409.056	95.058
5121	9	C01FE 870D2 13591	2009/042 10:45:37.568	406021754.6	-130.98	2409.056	95.058
5121	9	C01FE 870D2 13591	2009/042 10:46:29.471	406019120.0	-112.36	2409.056	95.058
5121	9	C01FE 870D2 13591	2009/042 10:47:21.373	406017641.4	-140.00	2409.056	95.058
5121	9	C01FE 870D2 13591	2009/042 10:47:21.373	406017641.4	-121.39	2409.056	95.058
5121	9	C01FE 870D2 13591	2009/042 10:48:10.144	406016883.8	-119.70	2409.056	95.058
5121	9	C01FE 870D2 13591	2009/042 10:48:58.361	406016457.0	-119.13	2409.056	95.058
5121	11	C01FE 870D2 13501	2009/042 11:18:12.954	406031582.1	-130.65	2399.750	83.363
5121	11	C01FE 870D2 13501	2009/042 11:19:53.594	406029664.2	-127.90	2399.750	83.363
5121	11	C01FE 870D2 13501	2009/042 11:21:34.254	406026314.5	-129.55	2399.750	83.363
5121	11	C01FE 870D2 13501	2009/042 11:22:22.494	406024355.8	-133.40	2399.750	83.363
5121	11	C01FE 870D2 13501	2009/042 11:23:10.714	406022465.0	-129.00	2399.750	83.363
5121	11	C01FE 870D2 13501	2009/042 11:24:01.054	406020783.8	-134.50	2399.750	83.363
5121	11	C01FE 870D2 13501	2009/042 11:24:51.374	406019489.0	-132.85	2399.750	83.363
5121	11	C01FE 870D2 13501	2009/042 11:25:43.274	406018523.9	-130.10	2399.914	95.275
5121	11	C01FE 870D2 13501	2009/042 11:26:32.034	406017884.3	-130.65	2399.914	95.275
5121	11	C01FE 870D2 13501	2009/042 11:26:32.034	406017884.3	-130.65	2399.582	80.116



N/A 9 N/A 8	N/A	6.099	53,359 51,431	14	2009/042 12:45:02 2009/042 17:01:32	2123B 802F1	9E8DD	NET	ħ	А	406 SARP 406 SARP	5121	32 33
N/A 8	N/A	37.404	55,979	4	2009/042 08:45:18			NET		А	ELT AC24	5121	34
N/A 9	N/A	N/A	N/A	2	2009/042 09:34:46	803BF 81FE0	1EDE4	NET !	Ŋ	A	TEST	5121	35
 N/A 11	N/A									A	406 SARP	5121	36
N/A 11	N/A									Α	406 SARP	5121	37.
N/A 10	N/A									Д	406 SARP	5121	38
13.067 9	46.000									Δ	406 SARP	5121	39
 N/A 10	N/A									Д	406 SARP	5121	40
N/A 10	N/A		N7	o for	ese two her	The				Α	TEST	5121	41
N/A 9	N/A		INZ	e ioi	ese two ner	110				Δ	406 SARP	5121	42
N/A 11	N/A									A	TEST	5121	43
N/A 10	N/A									A	TEST	5121	44
N/A 10	NA									E	ORB	5121	45
-1.639 9	48,114									4	TEST	5121	46



-36.9849,175.2092 - Google Maps

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				3	MOD IE	St 12 Ft	ebruary 2	.005				
										Po	wer radiat	ed
			ANT200	Primary A	ntenna		Antenna #	Desciption	BT100 test	121.5MHz	243MHz	406MH
Date	Location / Hex #	Test#	Time on	Time off	Time on	Time off]		Pov	ver Measure	ments	
12 February 2009	- 36.9849,175.2092	1	1000	1013	1013	1026	ANT110	Open	BT100 2008.54	82%	0	78%
	CO1ECC349DD40A1			1 - 2 - 2	3573	TETEN	Difference I		Vertical, dBuV/m	117.5	108.3	131.2
	C01ECC349DD40A1								Hoizont. dBuV/m	116.6	93.6	116.6
12 February 2009	- 36.9849,175.2092	1A	1000	1013	1013	1026	ANT110	Open	BT100 2008.55	81%	78%	75%
	CO1ECC349DD40A1				0	3			Vertical, dBuV/m	108.6	108.0	128.8
	C01ECC349DD40A1								Hoizont. dBuV/m	113.1	118.5	124.3
12 February 2009	- 36.9849,175.2092	2	1030	1035	1035	1049	1/4 Wave	Open	BT100 2008.58	64%	0	88%
	C01FE870D213591						32.00		Vertical, dBuV/m	100.3	85.3	115.5
	C01FE870D213591								Hoizont. dBuV/m	106.2	82.9	100.5
12 February 2009	- 36.9849,175.2092	3	1105	1114	1117	1135	1/4 Wave	Enclosed	BT100 2008.60	0	0	73%
	C01FE870D213501								Vertical, dBuV/m	69.9	78.5	97.8
	C01FE870D213501								Hoizont. dBuV/m	67.2	77.7	102.8
12 February 2009	- 36.9849,175.2092	4	1140	1145	1145	1200	ANT110	Enclosed	BT100 2008.62	22%	17%	62%
	C01FF870D2135D1								Vertical, dBuV/m	76.4	88.4	95.0
	C01FE870D2135D1						- 16		Hoizont. dBuV/m	76.5	92.2	111.0
# Antenna testea	l were both two frequent	cy										
antenna only apai	rt from ANT110 = 3 Frequ	uency						ANT200	BT100 2008.57	94%	77%	96%
								ANT200	BT100 2008.60	90%	81%	97%
est#1 Open ANT2	200 attached to simula	ite exter	ior anten	na								
est#2 Open SASD	activated with 3 Freq	ANT11) attached	to SASD								
est#3 Open SASD	activated with 2 Freq	1/4 wa	ve stubby	antenna a	attached t	o SASD						
est#4 Enclosed S	ASD activated with 2 F	req 1/4	wave stu	bby anten	na attache	ed to SASD						
est # 5 Enclosed S/	ASD activated with 2 F	req 1/4	wave stu	bby anten	na attache	ed to SASD						



SASD – SECONDARY ANTENNA SWITCHING DEVICE Provisional specification

Type Number: 15-56-0002		Units
Operating Frequencies	406.025/028	MHz
	121.5	MHz
λ	243.0	MHz
Insertion iss, input to main antenna port	0.25	dB
Insertion loss, input to alternative antenna port	0.25	dB
SWR all requencies	1.1:1	
Reverse isolation at 1 metre		dB
SWR detection level (adjustable)	3:1	
Activation delay first burst	5	ms
Activation delay subsequent bursts (locked)	0	ms
Power source - internat	9	volt
Current supply – not active	0	mA
Current supply - active	30	mA
Endurance - active	>24	Hrs
Battery expiry shelf life	5	years
Connector type all ports input /outpy El	BNC	female
Connector type back up antenna	TNC	mm
Antenna ¼ wave 406/121.5MHz		
Environmental standard to be approved	C126	TSO
Fail safe mode (electrical or battery fail) defaulte to connection	Main antenna.	

Dimensions SASD 170 x 60 x 35 mm (270grams)

Secondary Antenna 210mm long including TNC contector (53grams)

SASD Fitted preferably in line with ELT on the same structural member.



SASD with 1/4 wave antenna



SASD with 406AF-H ELT

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